United States of America
Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SR09240RC

This certificate issued to

Bell Helicopter Textron, Inc.
441 Industrial Park Road
Piney Flats, TN 37686

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 27 *of the* Federal Aviation *Regulations.*

Original Product -- Type Certificate Number : H2SW

Make : Bell Helicopter Textron

Model : 407

Description of Type Design Change:
Installation of FADEC Mode Switch Relocation in accordance with Premier Aviation, Inc. Master Drawing List F00-15003, Revision D, dated August 17, 2000, or later FAA approved revision.

Limitations and Conditions:

(See continuation sheet 3 of 3)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application : December 19, 1996

Date of issuance : March 16, 2000

Date reissued : April 15, 2011; January 17, 2002

Date amended :

By direction of the Administrator

(*Signature*)
James A. Richmond, Acting Manager
Rotorcraft Certification Office,
Southwest Region

(*Title*)

Any alteration of this certificate is punishable by a fine of not exceeding $1,000, or imprisonment not exceeding 3 years, or both.

 This certificate may be transferred in accordance with FAR 21.47.

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Supplemental Type Certificate

(Continuation Sheet)

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Supplemental Type Certificate

(Continuation Sheet)

Number SR09240RC

Date of Issuance: March 16, 2000
Date of Reissuance: April 15, 2011

Limitations and Conditions (Continued):

FAA approved Rotorcraft Flight Manual Supplement (RFMS) F09-62002 dated March 16, 2000, or later FAA approved RFMS is required.

Instructions for Continued Airworthiness, Premier Aviation, Inc., Document F08-72004, Revision - , dated May 11, 2000, or later FAA accepted revision is required for this installation.

If the helicopter configuration includes equipment(either previously installed or installed concurrently with this STC) for which a FADEC EMC Test is required (see BHTI Document 407-099-096) a FADEC EMC test is required after the installation of this STC.

The installer must determine whether this design change is compatible with previously approved modifications. The standard EMC test for intra-system compatibility between on-board electrical/electronic systems is the responsibility of the installer.

If the holder agrees to permit another person to use this certificate to alter a product, the holder must give the other person written evidence of that permission.